TM 55-1520-228-PMD C13

CHANGE

No. 13

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D. C., 18 July 1997

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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17 and 18	17 and 18
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TM 55-1520-228-PMD C 1 2

CHANGE

NO. 12

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 30 MARCH 1994

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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TM 55-1520-228-PMD C 1 1

CHANGE

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 13 April 1992

NO. 11

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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CHANGE

NO. 10

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OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE

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TM 55-1520-228-PMD C 9

CHANGE

NO. 9

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 19 February 1991

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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General United Army

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CHANGE

NO. 8

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OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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7 through 12	7 through 12
	12.1/12.2
17 and 18	17 and 18
18.1/18.2	18.1/18.2

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CHANGE

NO. 7

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WASHINGTON, D.C., 10 July 1989

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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TM 55-1520-228-PMD C 6

CHANGE

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 4 January 1988

NO. 6

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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This copy is a reprint of change 5.

URGENT

TM 55-1520-228-PMD C 5

CHANGE

NO. 5

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 10 November 1987

OH-58A/C HELICOPTERS
PREVENTIVE MAINTENANCE
DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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TM 55-1520-228-PMD

C 4

CHANGE

NO. 4

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 23 December 1986

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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11 through 14

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TM 55-1520-228-PMD C 3

CHANGE

NO. 3

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 17 May 1985

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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TM	55-1520-228-PMD
	C 3

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TM 55-1520-228-PMD C 2

CHANGE

No. 2

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 30 March 1984

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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To be distributed in accordance with DA Form 12-31, PM Maintenance Requirements for OH-58 and OH-58C aircraft.

TM 55-1520-228-PMD C 1

CHANGE

No. 1

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 8 March 1982

01-I-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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DISTRIBUTION:

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HEADQUARTERS
DEPARTMENT OF THE ARMY
Washington, D. C., 8 August 1980

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

GENERAL INFORMATION AND SCOPE

WARNING: CERTAIN INSPECTIONS ARE MANDATORY SAFETY-OF-FLIGHT REQUIREMENTS, AND THE INSPECTION INTERVALS CANNOT BE EXCEEDED. IN THE EVENT THESE INSPECTIONS CANNOT BE ACCOMPLISHED AT DAILY INTERVAL, THE AIRCRAFT CONDITION STATUS SYMBOL WILL BE IMMEDIATELY CHANGED TO A RED X. THESE TYPE INSPECTION ITEMS ARE PRECEDED BY "MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEM".

NOTE: INDIVIDUAL INSPECTION ITEMS CONTAINED IN THIS MANUAL ARE CONSIDERED THE MINIMUM REQUIREMENTS FOR PERFORMING A DAILY INSPECTION AND MUST BE PERFORMED, THE CUMULATIVE EFFECTS OF INSPECTION DEFERRALS ARE UNKNOWN AND COULD RESULT IN CATASTROPHIC FAILURE OR INCREASED MAINTENANCE AT A LATER DATE. THEREFORE, THE USE OF SPECIAL LETTERING TO EMPHASIZE CRITICAL INSPECTION IS NOT TO BE CONSTRUED AS AUTHORITY FOR DEFERRAL OF OTHER INSPECTIONS.

^{*}This manual together with TM 55-1520-228-PM, 1 September 1978, supersedes TM 55-1520-228-PMS, dated 24 September 1976, including all changes.

- 1. Inspection Requirements. This manual contains complete requirements for daily inspection for OH-58 helicopters. It does not contain instructions for repair, adjustment, or other means of rectifying conditions, nor does it contain instructions for troubleshooting to find causes for malfunctioning. Specific tolerances, limits, etc., can be found in the applicable maintenance manuals. Use of the alphabetical index in the applicable manuals will facilitate locating the required information.
- **2. Maintenance Activities.** The inspections prescribed by this manual will be performed at specified periods by Aviation Unit Maintenance (AVUM) activities with assistance of Aviation Intermediate Maintenance (AVIM) and Depot activities when required.

3. General Information.

- a. The inspection requirements contained herein are stated in such a manner as to establish what and when certain equipment is to be inspected and what conditions are desired/undesired. Compliance with the provisions outlined herein is required in order to assure that proper servicing has been accomplished and latent defects are discovered and corrected before malfunctioning or serious trouble results. Inspection requirements are arranged, as nearly as possible, according to the manner in which they will be performed. The requirements are divided into groups under area headings (figure 1).
- **b.** The inspection intervals designated herein will not be exceeded except in actual operational emergencies as ex-

- plained herein. It is the commander's responsibility to determine (on an individual aircraft basis) when inspection intervals may be exceeded. For this purpose, operational emergencies are conditions of combat, or conditions of disaster which necessitate flight to evacuate aircraft or personnel. When aircraft are operated beyond the normal inspection due-time because of such emergency situations, a circled red X status symbol and an appropriate statement (to include authority) must be entered in blocks 16 and 17 of DA Form 2408-13 (Aircraft Inspection and Maintenance Record) until such time as the inspection is complete. Since safety may be jeopardized when inspections are delayed to meet emergency requirements, commander will ensure that the aircraft status symbol reverts to a red "X" and that delayed inspections are accomplished immediately upon termination of the actual emergency. When unusual local conditions of environment, utilization, mission, experience of flight crew and maintenance personnel, periods of inactivity, etc., are encountered, the maintenance officer will, at his discretion, increase the scope and/or frequency of maintenance or inspections as necessary to ensure safe flight.
- **c.** This manual may contain inspection requirements applicable to specific equipment not installed on your aircraft. Those requirements should be disregarded.
- **d.** DA Form 2408-13 will be used to record all deficiencies or shortcomings discovered during the inspection.
- e. A 1-1/2 inch space between each area of inspection is being provided to allow insertion of additional inspection items as required by local command inspection procedures.

4. Special Instructions.

- a. A Preventive Maintenance Daily inspection is accomplished after the last flight of the day, or prior to the first flight on the next day on which the aircraft is flown. The inspection consists of visual examination and operational checks to determine that the aircraft can safely and efficiently perform the assigned mission.
- b. Work time requirements to accomplish each inspection are stated at top of the checklist.

5. Reporting Errors and Recommending Improvements.

You can help improve this manual. If you find any mistake

or if you know of a way to improve the procedures, please let us know. Mail your letter, DA Form 2028 (Recommended Changes to Publications and Blank Forms) or DA Form 2028-2 located in the back of the applicable aircraft maintenance manual (when using the 2028-2 from the maintenance manual, insure that the publication number and title refer to this PMD) directly to Commander, US Army Troop Support and Aviation Materiel Readiness Command, ATTN: DRST-MTPS(1), 4300 Goodfellow Blvd. St. Louis, MO 63120. A reply will be furnished to you.

6. Inspection Areas. Inspection areas are shown in figure 1.

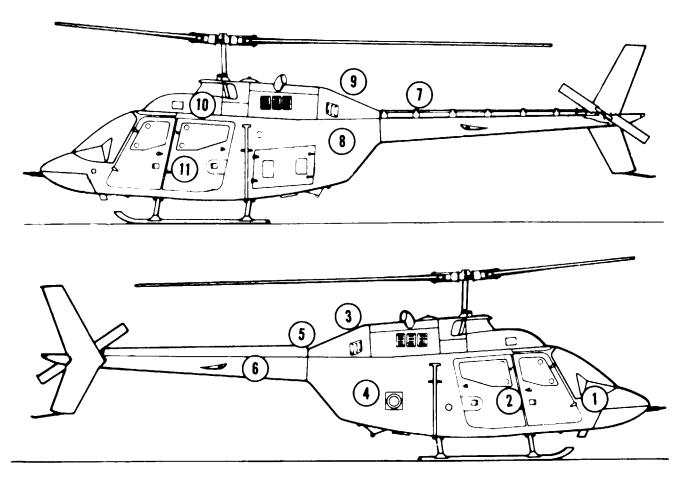


Figure 1. Inspection Areas.

Area No. 1	Nose Area	All surfaces and components in nose compartment and on exterior ahead of crew doors.
Area No. 2	Cabin Electrical Equipment Shelf and Landing Gear Area (Right Side)	All surfaces, components, and equipment inside cabin, and on cabin exterior between forward side of crew doors and aft side of passenger doors and cabin overhead. Includes complete landing gear and fuel cell sumps and filler. All the equipment in the area of the aft electrical equipment shelf,
Area No. 3	Engine Area (Right Side)	All surfaces, components, and equipment associated with engine installation, located above engine work deck.
Area No. 4	Aft Fuselage (Right Side)	All surfaces, components, below engine deck level, between cabin area and tail boom attachment bulkhead.
Area No. 5	Aft Fairing and Oil Cooler (Right Side)	All surface aft of engine cowling and oil tank.
Area No. 6	Tail Boom	All surfaces, components, and equipment located in the tail boom and vertical fin structure. Includes tail rotor, horizontal stabilizer, and control linkages. Also all sup- ports, bearings, and shafting mounted on tail boom.

Area No. 7	Aft Fairing and Oil Cooler (Left Side)	All surface aft of engine cowling, tail rotor drive shaft and oil cooler blower.
Area No. 8	Avionics and Aft Fuselage Area (Left Side)	All surfaces, components, and equipment in fuselage below engine deck level, between cabin area and tail boom attachment bulkhead.
Area No. 9	Engine Area (Left Side)	All surfaces, components, and equipment associated with engine installation, located above engine work deck.
Area No. 10	Transmission and Pylon Area	All surfaces, components, and equipment of the main rotor pylon group, from top mast to cabin roof. Includes main rotor, mast and rotating controls, transmission with accessories and mounts, servo actuators, and hydraulic system.
Area No. 11	Cabin and Landing Gear Area (Left Side)	All surfaces, components, and equipment inside cabin, and on cabin exterior between forward side of crew doors and aft side of passenger doors and cabin overhead. Includes complete landing gear.

PREVENTIVE MAINTENANCE DAILY CHECKLIST

The Preventive Maintenance Daily Checklist will be accomplished following the last flight of the day or prior to the first flight on the next day on which the aircraft is flown. The inspection consists of visual examination and operational checks to determine that the aircraft can safely and efficiently perfoom its assigned mission.

DAILY INSPECTION TOTAL WORK TIME: 2.3 Work Hours

Seq. No.	Item and Procodure	C8	Seq No.	Item and Proc.dure
	NOSE AREA No. 1		1.5	Radar warning antennas for damage and security.
1.1	Inspect aircraft forms and records for recorded discrepancies (DA PAM 738-751).		1.6	Landing lights, brackets, terminals and wiring for condition.
1.2	Nose section exterior for visible damage.		1.7	Landing/search light for damage and security.
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		1.8	UHF antenna for condition and security
	INSPECTION ITEM		1.9	Vent system drain for obstructions, condition.
1.3	Pitot tube and static ports for obstructions and cleanliness.		1.10	Chin bubbles for damage and cleanliness.
1.4	Forward proximity warning system antenna for condition and security.		1.11	Windshields and windows for damage and cleanliness.

"FOD REMINDER"

Check work area for tools and parts after completion of maintenance and inspection.

Seq. No.	Item and Procedure	C2	Seq. No.	Item and Procedure
1.12 1.13	WSPS Windshield Deflector for damage and security. WSPS Lower Cutter Assembly for damage and security.		2.8	Crew and passenger doors for positive latching and proper operation, windows and vents for cleanliness and damage. Check hinges and con- dition of weather stripping. Door posts for
1.14	Breakaway Tip for looseness or play.			cracks.
	CABIN AND LANDING GEAR AREA No.2 (RIGHT SIDE)		2.9	Crew and passenger door jettison handles for proper installation and breakaway safetywire.
2.1	Cabin exterior for damage.		2.10	Cabin interior for cleanliness, proper stowage of
2.2	Check fuel for water and other contamination. Drain fuel sump (use sample jar) through the sump drain.		2.11	equipment and visible damage. Nose section interior for cleanliness; equipment for visible damage and loose connections.
2.3	Landing gear for condition of skid tubes, cross tubes, attachment fittings and skid shoes for evidence of hard landing damage and security.			MANDATORY SAFETY OF FLIGHT
2.4	Radar altimeter antennas for damage and			INSPECTION ITEM
2.5	security. Transponder antenna for damage and security.		2.12	Anti-torque pedals for freedom of operation and security.
2.6	Radar warning antenna for damage and security (blade).		2.13	Current compass correction card for
2.7	Marker beacon antenna for damage and security.			availability and-legibility.

Seq. No.	Item and Procedure	Seq. No.	Item and Procedure
2.14	Instrument panel and instruments for security, condition and range markings.		MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
2.15	FAT gage for condition and security.	2.22	Cyclic control for freedom of operation.
2.16	Check fuel valve handle-on (FWD) and off (AFT) [hen on (FWD) for binding, proper operation, condition, and security.	2.23	Cyclic, boots and grips for condition, security.
2.17	Overhead console for condition and security.	2.24	Copilot cyclic control for security of electrical connectors.
2.18	Ulility light and wire for condition security, and proper mounting.	2.25	Seat bottom for security, damage, and condition of web- bing and date of installation. Seat back cushions for cleanliness and condition.
2.19	Heater control for condition and security. MANDATOR SAFETY OF FLIGHT INSPECTION ITEM	2.26	Safety belts and shoulder harness for damage, corrosion, cuts, fraying, and security. Inertia reels for damage, security, and positive locking and unlocking. Safety belt mounting hinges and brackets for cracks (visual). Check
2.20	Collective and power controls for freedom of operation.	2.27	retarder springs for proper operation. Armor panels for condition and security.
2.21	Collective boots for condition, security and throttle set screw.		

"FOD REMINDER" CHECK WORK AREA FOR TOOLS AND PARTS AFTER COMPLETION OF MAINTENANCE AND INSPECTION.

C12

Seq. No.	Item and Procedure	Seq. No.	Item and Procedure
	Fire extinguisher for designated location, presence of inspection date tag, broken or missing seal, pressure indicator in green, extinguisher and brackets secure. First aid kits for designated location, presence of inspection date tag, broken or missing seal, completeness of side pocket contents, legible identification markings, and security (refer to TM 55-1500-204-25/1). Passenger seat and back cushions, retainer loops and tabs for damage, condition and security. Sound-proofing for condition, security, and cleanliness. Aft electrical equipment shelf for unauthorized or foreign objects, cleanliness, and security of components.		ENGINE AREA No. 3 (RIGHT SIDE) MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Engine air inlet bellmouth assembly through engine inlet fairing window for separation, cracking, deformation, security and obstructions and plenum area for loose or foreign objects. Particle separator for damage and particle ejection opening for freedom of obstructions, and loose vortex tubes (generators). Inspect particle separator to induction fairing for fit and sealing. Firewall to induction fairing for fit and sealing. Engine cowling and fairing for damage, security, and condition of fasteners and hinges.
		3.4	Anti-collision light for condition and security; loose connections.
		3.5	Engine mounts and engine mount fittings, for cracks, damage and security.
		3.5.1	Remove chip detector on freewheeling assembly and check for metal particles and reinstall.
		3.6	Fuel control linkage for damage and security.

"FOD REMINDER" CHECK WORK AREA FOR TOOLS AND PARTS AFTER COMPLETION OF MAINTENANCE AND INSPECTION.

		1 60 1		.
Seq. No.	Item and Procedure.	C8	Seq No	Item and Procedure
3.7	Compressor armor for condition and security.		3.12	Starter generator for condition. security Exhaust duct for condition, obstruction, and security
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		3.13	Main input driveshaft coupling for evidence of overheating. Exterior plating for heat discoloration or blistering. Paint strips (if used) for discoloration.
3.8	Compressor, bleed valve for damage and obstruction. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM			MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
3.9	Bleed air elbows (2) for damage.			Inspect main rotor driveshaft 206-040-371 for damage and fastener security.
3.10	Engine, accessories. and connections for damage and security.			AFT FUESELAGE AREA No. 4 (RIGHT SIDE)
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		4.1	Fuel filler cap for proper locking and condition of seal, security of lanyard.
3.11	Exhaust stacks for damage, corrosion obstructions. and security, Exhaust stack clamp for cracks, corrosion. and security.		4.1.1	If installed, inspect rubber pad below fuel receiver for debonding.

Seq No.	ltem and Procedure	C8	Seq. No.	Item and Procedure
4.2	FM homing antenna for condition and security.			
4.3	Battery and fuel vents for condition, obstruction, proper positioning and security.			
4.4	Air ducts for condition and security.			
4.5	External power access door for condition and security.			
4.6	Radar warning antennas for damage and security.			

"FOD REMINDER"

Seq. No.	Item and Procedure	Seq. No.	Item and Procedure
	AFT FAIRING AND OIL COOLER AREA No. 5 (RIGHT SIDE) MANDATORY SAFETY		age, bonded fittings for security, splined adapters for freedom of movement. Do not wipe grease from seal area. Inspection not required after compliance with MWO 55-1520-228-50-25.
		6.3.1	Check segmented driveshaft for condition, shaft coupling disc for distortion, cracks and security, splined
5.1	Engine oil tank for damage, security and servicing lines for leaks and damage. Loose or missing identification tape on oil lines.		adapter from freedom of movement, and bearing and driveshaft for evidence of over heating. Do not wipe grease from seal area. Required after compliance with MWO 55-1520-228-50-25.
5.2	Oil filler access door for condition and security.	6.4	Horizontal stabilizer for visible damage.
5.3	Engine external scavenge oil filter for damage, security, lines for leaks and damage, and bypass indicator (red but-	6.5	Navigation lights for condition and security.
	ton) in. Loose or missing identification tape on oil lines. After compliance with MWO 55-1520-228-50-44. TAIL BOOM AREA No. 6 MANDATORY SAFETY OF FLIGHT INSPECTION ITEM	6.6	VOR antennas for damage and security.
		6.7	Tail light and support for damage and security.
		6.8	Vertical stabilizer, tail skid, and antenna leads for security and damage.
		6.9	Ballast installation for security and condition.
6.1	Tail boom exterior for visible damage.		MANDATORY SAFETY OF FLIGHT
6.2	Driveshaft cover for damage, security, and condition of fasteners.	6.10	INSPECTION ITEM Tail nature controls for demage and security of attach
6.3	Long tail rotor drive shaft and bearings for evidence of overheat; inspect slippage marks, excessive grease leak-	0.10	Tail rotor controls for damage and security of attach- ment bolts.

"FOD REMINDER" CHECK WORK AREA FOR TOOLS AND PARTS AFTER COMPLETION OF MAINTENANCE AND INSPECTION. 12.1/(12,2 blank)

Seq. No.	Item and Procedure	C10	Seq. No.	Item and Procedure
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM			MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
6.10.1 6.11 6.12 6.12.1 6.12.2	Tail Rotor Pitch Links For cracks using ten-power magnifications. Pay particular attention to area around ball bearing at each end of Pitch Link. Suspected cracks shall be confirmed by penetrant inspection. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Tail rotor gear box for security, oil level, condition, and vent for obstruction. Check sight glass for cracks, damage, and stains that might give a false indication of the oil level. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Tail rotor hub and blade assembly for security and damage. Tail rotor blades for cleanliness and clean as required. Inspect all self-locking nuts on the tail rotor blade and hub assembly for cracks, damage and security.		6.14 6.15 6.16 6.17	Inspect both tail rotor blades for cracks on both sides in the area approximately 7 inches outboard from the butt end and 1.5 inches aft of the leading edge. If inspection reveals a crack, blade must be removed for evaluation prior to next flight. Tail rotor blades with serial number TLL-8000 and below require ten-power magnification for this inspection and careful inspection for corrosion and deterioration of clear coating on areas left clear for inspection. Not required after compliance with MWO 55-1520-228-50-25. Vulnerability reduction control system access panel for damage and security. VOR antenna for damage and security. Horizontal stabilizer for visible damage. Navigation lights for condition and security. MANDATORY SAFETY OF FLIGHT Inspect tail rotor blade tip blocks for cracks, debonding, loose, missing or corroded rivets.

13

Seq No.	Item and Procedure	C2	Seq. No.	Item and Procedure
	AFT FAIRING AND OIL COOLER AREA No. 7 (LEFT SIDE) MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		8.2 8.3 8.4 8.5	ADF sense antenna for damage and security. ADF loop antenna for damage and security. Avionics compartment door for condition. security, proper operation. Avionics compartment for . loose or
7.1	Engine oil tank for damage, security and servic- ing, lines for leaks and damags.		8.6	unauthorized equipment. Foam pad for condition, cleanliness, and security.
7.2	Oil cooler blower and drive shaft for damage, cracks, and security. Blower air intake for foreign material and clogging.		8.7	Battery for security and leakage.
7.3	Aft fairing assembly for damage and condition of fasteners.		8.8	External power receptacle for condition and security. FM homing antennas for damage and security.
	AVIONICS AND AFT No. 8 FUSLAGE AREA (LEFT SIDE)			
8.1	Aft proximity warning system antenna for damage and security.			

C12

Seq. No.	Item and Procedure	Seq. No.	Item and Procedure
110.	ENGINE AREA No, 9 (LEFT SIDE)	9.8	Linear actuator, linkage, and electrical lead for conditionand security.
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM	9.9	Engine automatic relight for condition and security.
).1	Engine air inlet bellmouth assembly through engine inlet fairing window for separation, cracking, deformation, security and obstructions and plenum area for loose or	9.10	Exhaust stacks for damage, corrosion, obstructions, an security. Exhaust stack clamp for cracks, corrosion, an security.
0.2	foreign objects. Particle separator for damage and particle ejection open-	9.11	Engine combustion chamber housing, turbine support assembly and exhaust ducts for cracks, dents, burned, obuckled areas.
).3	ing for freedom of obstructions. Engine cowling and fairing for damage, security, and condition of fasteners and hinges.	9.12	Drain valve for leakage, security and contact with driv shaft.
.4	Engine mounts and engine mount fittings, for cracks, damage and security,		MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
.5	Compressor armor for condition and security.	9.13	Fuel nozzle and lines for security and leaks.
0.6	Airframe mounted fuel filter for condition, leaks and sc- curily. After compliance with MWO 1-1520-228-50-48.	9.14	Forward tail rotor driveshaft for cleanliness, corrosion and freedom of movement on the splined adapters.
).7	Engine, accessories, and connections for damage and security.		

"FOD REMINDER" CHECK WORK AREA FOR TOOLS AND PARTS AFTER COMPLETION OF MAINTENANCE AND INSPECTION. TM 55-1520-228-PMD

C3

Seq.	I] [Seq.	
No.	Item and Procedure		No.	Item and Procedure
	TRANSMISSION AND PYLON AREA No. 10 (LEFT SIDE) MANDATORY SAFETY OF FLIGHT INSPECTION ITEM			surfaces. If inspection by flashlight reveals a crack indication, validate the crack indication using fluorescent penetrant. If crack is confirmed, remove yoke assembly from service. MANDATORY SAFETY OF FLIGHT
10.1	Main rotor hub assembly for leakage, damage, security, and reservoirs for proper servicing.		10.6	INSPECTION ITEM Main rotor blades for condition, damage, and security.
10.2 10.3 10.4	Sight glass for damage and stains which might give false oil level indication. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Check pillow blocks for mounting condition, security of mounting bolts and slippage marks. If slippage has occurred, replace hardware. Static stops for damage and security. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Main rotor yoke assembly for cracks (visual) between the pillow block bores and lower flange on inboard surfaces. Wipe clean yoke web section inner and outer		10.6.1 10.7 10.8	Main rotor blades for cleanliness and clean as required. Main rotor blades for corrosion, specifically in the area of the inertia weight retention screw holes. Tip cap firing for condition and security. Check main rolor bladdes for protective coating, clean blades and re-wax as required. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Visible areas of mast for damage.

"FOD REMINDER" CHECK WORK AREA FOR TOOLS AND PARTS AFTER COMPLETION OF MAINTENANCE AND INSPECTION.

Item and Procedure	C9	Seq. No.	Item and Procedure
Boot for damage and security.	Ī		MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Swashplate, lever and sleeve, and connecting linkage		10.12.1	Main rotor pitch links for damage and security of attachment, tubes for corrosion and cracks; pay particular attention to swagged ends at jam nut.
base drain holes open and free of foreign matter.		10.13	Hydraulic servo actuator support assembly for cracks, corrosion, security and damage.
NOTE Ensure swashplate is level for inspection of outerring self-aligning bearings for main rotor push-pull tubes.		10.14	Hydraulic system components and lines for security, chafing, damage, leaks and reservoir for servicing. Collective actuator bellcrank cover for missing rubber flipper or flipper bond failure. Proper security of hydraulic reservoir cap and pin assembly. (Refer to TM 55-1520-228-23-1).
MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		10.15	Hydraulic filter element indicator for filter clogged indication.
Cyclic and collective control linkage for security and condition. Check for security of attachment bolts by attempting to torque nuts with fingers (DO NOT REMOVE COTTER PIN).		10.16	Edges of main drive shaft cover assembly for evidence of grease leakage. Security of attaching hardware.
	Boot for damage and security. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Swashplate, lever and sleeve, and connecting linkage for security and visible damage. Swashplate support base drain holes open and free of foreign matter. NOTE Ensure swashplate is level for inspection of outerring self-aligning bearings for main rotor push-pull tubes. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Cyclic and collective control linkage for security and condition. Check for security of attachment bolts by attempting to torque nuts with fingers (DO NOT)	Boot for damage and security. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Swashplate, lever and sleeve, and connecting linkage for security and visible damage. Swashplate support base drain holes open and free of foreign matter. NOTE Ensure swashplate is level for inspection of outerring self-aligning bearings for main rotor push-pull tubes. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Cyclic and collective control linkage for security and condition. Check for security of attachment bolts by attempting to torque nuts with fingers (DO NOT)	Boot for damage and security. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Swashplate, lever and sleeve, and connecting linkage for security and visible damage. Swashplate support base drain holes open and free of foreign matter. NOTE Ensure swashplate is level for inspection of outerring self-aligning bearings for main rotor push-pull tubes. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Cyclic and collective control linkage for security and condition. Check for security of attachment bolts by attempting to torque nuts with fingers (DO NOT)

Seq. No.	Item and Procedure	C13	Seq. No.	Item and Procedure
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		10.22	Pylon isolation mount cover for condition, drag pin assembly to static stop or cabin roof for foreign objects and damage.
10.17	Power turbine governor (N2) linkage for freedom of operation, security, and condition.		10.22.1	Wipe up oil spillage and clean area using cloth dampened with dry cleaning solvent. Wipe dry with clean cloth. Visually examine clevis area and mount
10.18	Sight glass for damage or stains which might give false oil level indication.			for cracks, nicks, scratches, and loose or missing screws, bolts, and cotter pins.
10.19	Rotor tachometer generator, hydraulic pump, for condition and security.		10.23	Inspect area under drag pin bearing for oil and water.
10.20	Transmission oil cooler and duct for condition, security and obstruction.		10.24	Inspect transmission deck area for dents, cracks, holes, and overall
10.21	Transmission oil filter and filter head for condition, security, and leaks.			condition.
10.21.1	Roof mounted transmission oil filter for damage, security, lines for leaks and		10.25	Pylon support links for cracks, condition (visual) and security of hardware.
	damage, and bypass indicator (red button) in. After compliance with MWO 1-1520-228-50-51.		10.26	Inspect Electrical wiring for damage, security, and corrosion.

Seq. No.	Item and Procedure	C 13	Seq. No.	Item and Procedure
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		10-33	FM No. 2 antenna for damage and security.
10.27	Transmission and connections for damage and security. Transmission for		10-34	WSPS Upper Cutter Assembly for damage and security.
10.28	servicing. (Filler cap for security.) Particle separator for damage and particle swirls for condition and security.		10-35	GPS antenna mounted on WSPS upper cutter assembly for damage and security. After compliance with MWO 1-1520-228-50-53.
10.29	Check security of engine cleaning provision tube and cap assembly.			
10.30	Transmission faking for damage and condition of fasteners.			
10.31	IFF antenna for damage and security.			
10.32	Glide slope antenna for damage and security.			
			l	l l

Seq. No.	Item and Procedure		Seq. No.	Item and Procedure
	CABIN AND LANDING			POWER ON
	GEAR AREA No. 11 (LEFT SIDE)		1.15	Pitot heater for operation.
11.1	Landing gear for condition of skid tubes, cross tubes, attachment fittings and skid shoes for		1.16	Warning and caution panel lights for illumination on test.
11.2	evidence of hard landing damage and security. Crew and passenger doors for positive latching		2.33	Interior and instrument panel lights for proper operation.
	and proper operation, windows and vents for cleanliness and damage. Check hinges and condition of weather stripping.			WARNING
11.3	Seats for security, damage, and condition of webbing. Cushions for cleanliness and condition and installation date.			Assure defogging blower motors are free of foreign debris prior to energizing motors.
11.4	Armor plating for condition and security.		2.34	Operate the defogging blower motors for a minimum of 15 seconds to insure proper func-
11.5	Crew and passenger door jettison handles for proper installation and breakaway safety wire.		All Areas	tioning. Exterior lights for proper operation.
	LUBRICATION			
All Areas	Service in accordance with TM 55-1520-228-23, applicable to the daily requirements.			

By Order of the Secretary of the Army:

E. C. MEYER

General, United States Army

Chief of Staff

Official:

J. C. PENNINGTON

Major General, United States Army

The Adjutant General

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The Metric System and Equivalents

Linear Measure

1 centimeter = 10 millimeters = .39 inch 1 decimeter = 10 centimeters = 3.94 inches 1 meter = 10 decimeters = 39.37 inches 1 dekameter = 10 meters = 32.8 feet 1 hectometer = 10 dekameters = 328.08 feet 1 kilometer = 10 hectometers = 3,280.8 feet

Weights

1 centigram = 10 milligrams = .15 grain 1 decigram = 10 centigrams = 1.54 grains 1 gram = 10 decigram = .035 ounce 1 dekagram = 10 grams = .35 ounce 1 hectogram = 10 dekagrams = 3.52 ounces 1 kilogram = 10 hectograms = 2.2 pounds 1 quintal = 100 kilograms = 220.46 pounds 1 metric ton = 10 quintals = 1.1 short tons

Liquid Measure

1 centiliter = 10 milliters = .34 fl. ounce 1 deciliter = 10 centiliters = 3.38 fl. ounces 1 liter = 10 deciliters = 33.81 fl. ounces 1 dekaliter = 10 liters = 2.64 gallons 1 hectoliter = 10 dekaliters = 26.42 gallons 1 kiloliter = 10 hectoliters = 264.18 gallons

Square Measure

1 sq. centimeter = 100 sq. millimeters = .155 sq. inch 1 sq. decimeter = 100 sq. centimeters = 15.5 sq. inches 1 sq. meter (centare) = 100 sq. decimeters = 10.76 sq. feet 1 sq. dekameter (are) = 100 sq. meters = 1,076.4 sq. feet 1 sq. hectometer (hectare) = 100 sq. dekameters = 2.47 acres 1 sq. kilometer = 100 sq. hectometers = .386 sq. mile

Cubic Measure

1 cu. centimeter = 1000 cu. millimeters = .06 cu. inch 1 cu. decimeter = 1000 cu. centimeters = 61.02 cu. inches 1 cu. meter = 1000 cu. decimeters = 35.31 cu. feet

Approximate Conversion Factors

To change	To	Multiply by	To change	To	Multiply by
inches	centimeters	2.540	ounce-inches	newton-meters	.007062
feet	meters	.305	centimeters	inches	.394
yards	meters	.914	meters	feet	3.280
miles	kilometers	1.609	meters	yards	1.094
square inches	square centimeters	6.451	kilometers	miles	.621
square feet	square meters	.093	square centimeters	square inches	.155
square yards	square meters	.836	square meters	square feet	10.764
square miles	square kilometers	2.590	square meters	square yards	1.196
acres	square hectometers	.405	square kilometers	square miles	.386
cubic feet .	cubic meters	.028	square hectometers	acres	2.471
cubic yards	cubic meters	.765	cubic meters	cubic feet	35.315
fluid ounces	milliliters	29.57 3	cubic meters	cubic yards	1.308
pints	liters	.473	milliliters	fluid ounces	.034
quarts	liters	.946	liters	pints	2.113
gallons	liters	3.785	liters	quarts	1.057
ounces	grams	28.349	liters	galions	.264
pounds	kilograms	.454	grams	ounces	.035
short tons	metric tons	.907	kilograms	pounds	2.205
pound-feet	newton-meters	1.356	metric tons	short tons	1.102
pound-inches	newton-meters	.11296	men ie vons	011014 90119	

Temperature (Exact)

°F	Fahrenheit		
	temperature		

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